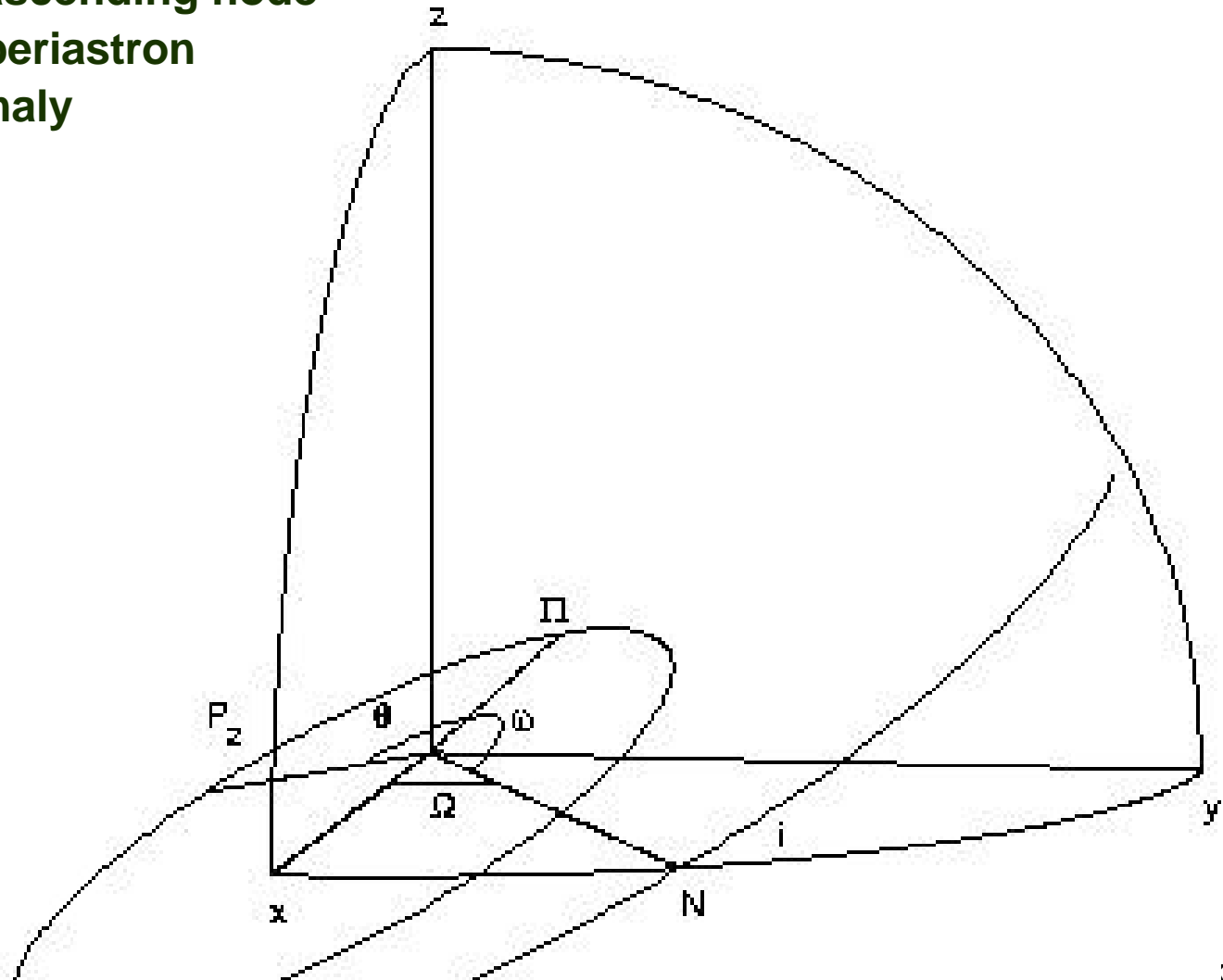
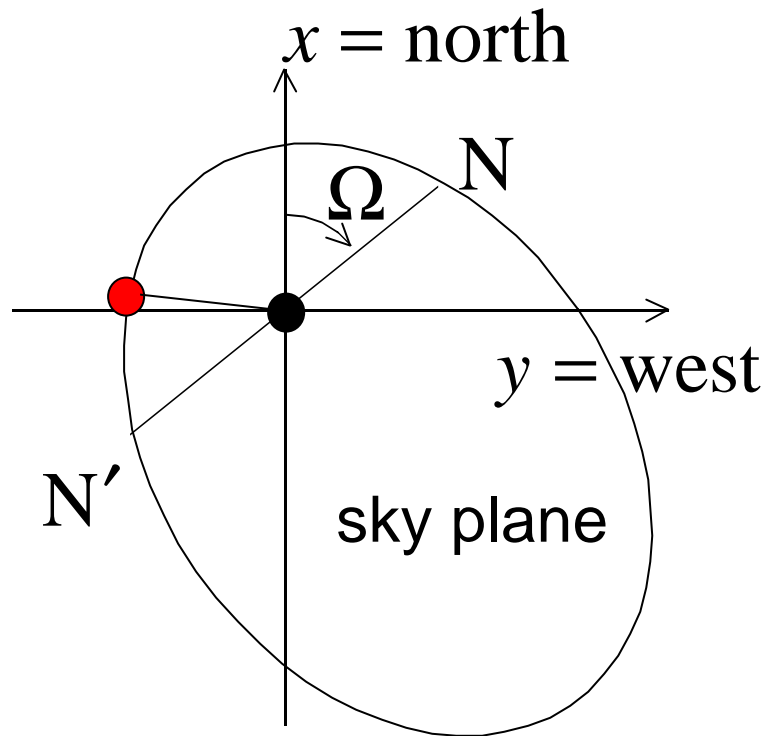


Orbit in Space

- x, y = north, west on sky plane thru m_1 at $x, y = 0$, observer at $-z$.
- i = inclination between sky and orbit planes
- N = ascending node, m_2 crosses x - y plane in $+z$ direction
- W** = longitude of ascending node
- w** = longitude of periastron
- θ = true anomaly



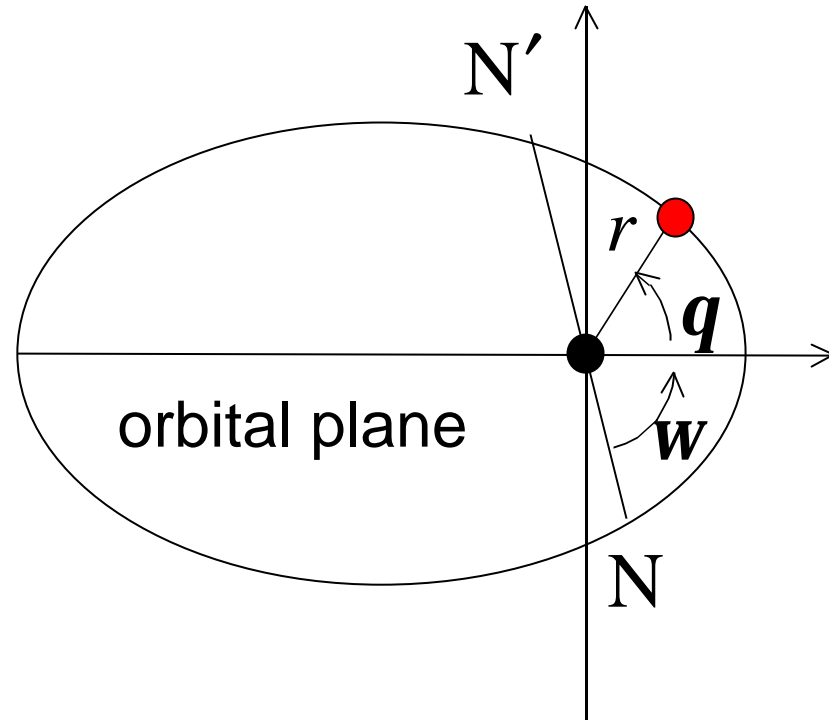
Relative orbit on the Sky



$$x = x' \cos \Omega + y' \sin \Omega \cos i$$

$$y = x' \sin \Omega - y' \cos \Omega \cos i$$

$$z = y' \sin i$$



$$x' = r \cos(\mathbf{q} + \mathbf{w})$$

$$y' = r \sin(\mathbf{q} + \mathbf{w})$$

Orbital Elements

specific angular momentum :

$$L_x = L \sin i \sin \Omega$$

$$L_y = -L \sin i \cos \Omega$$

$$L_z = L \cos i$$

7 orbital elements :

size (a), shape (e),

orientation in space (i, ω, Ω)

and in time (T, P).

Spectroscopic Binaries

- **m2 at (r, $\mathbf{q} + \mathbf{w}$)**

- project along the line of nodes: $r \cos(\mathbf{q} + \mathbf{w})$
- perpendicular to line of nodes: $r \sin(\mathbf{q} + \mathbf{w})$
 - project along line of sight: $z = r \sin(\mathbf{q} + \mathbf{w}) \sin i$

- **radial velocities along line-of sight**

$$V_{rad} = \dot{z} = \sin i \left[\dot{r} \sin(\mathbf{q} + \mathbf{w}) + r \dot{\mathbf{q}} \cos(\mathbf{q} + \mathbf{w}) \right]$$

use

$$r = \frac{a(1 - e^2)}{1 + e \cos \mathbf{q}} \quad \rightarrow \quad \dot{r} = \frac{e \sin(\mathbf{q}) r \dot{\mathbf{q}}}{1 + e \cos \mathbf{q}}$$

and Kepler's 2nd Law

$$r^2 \dot{\mathbf{q}} = \frac{2p a^2 (1 - e^2)^{1/2}}{P}$$

$$V_{rad} = \frac{2p a \sin i}{P \sqrt{1 - e^2}} \left[\cos(\mathbf{q} + \mathbf{w}) + e \cos(\mathbf{w}) \right]$$

Spectroscopic Orbital Velocities

$$V_{rad} = K [\cos(\mathbf{q} + \mathbf{w}) + e \cos \mathbf{w}] + \mathbf{g}$$

$$K = \frac{2\mathbf{p} a \sin i}{P\sqrt{1-e^2}}$$

ascending node: $\cos(\mathbf{q} + \mathbf{w}) = +1$

$$V_{\max} = K [e \cos \mathbf{w} + 1] + \mathbf{g}$$

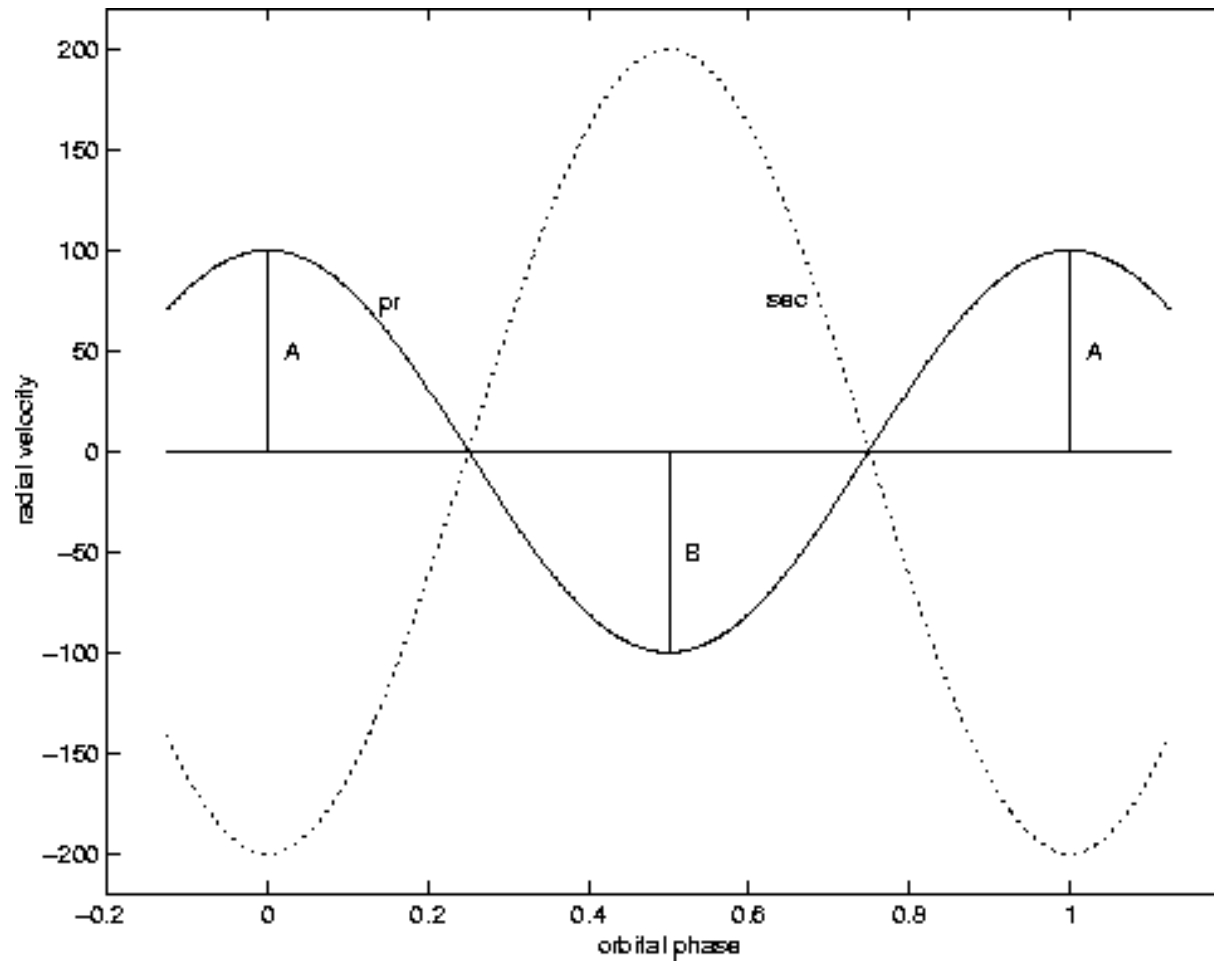
descending node: $\cos(\mathbf{q} + \mathbf{w}) = -1$

$$V_{\min} = K [e \cos \mathbf{w} - 1] + \mathbf{g}$$

semi-amplitude:

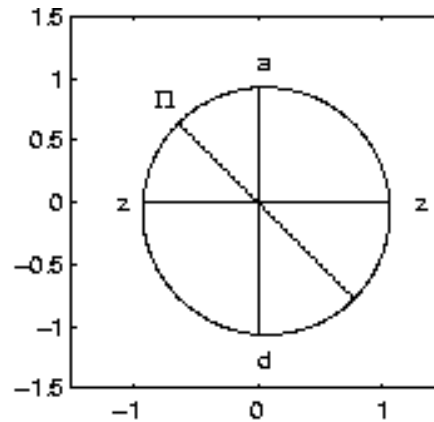
$$K \equiv (V_{\max} - V_{\min}) / 2$$

Circular Orbit

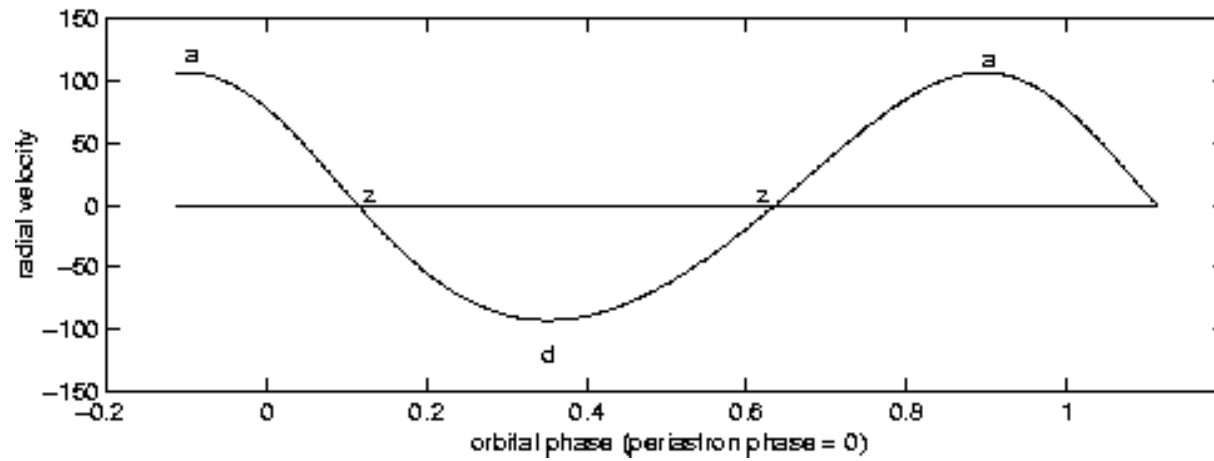


- $K_1 = 100 \text{ km/s}$ $K_2 = 200 \text{ km/s}$ $q = m_2 / m_1 = 0.5$

Radial Velocities

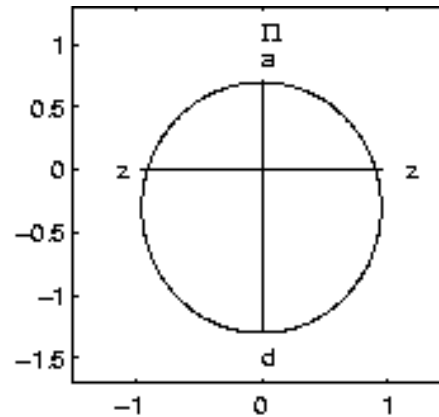


to the observer -->

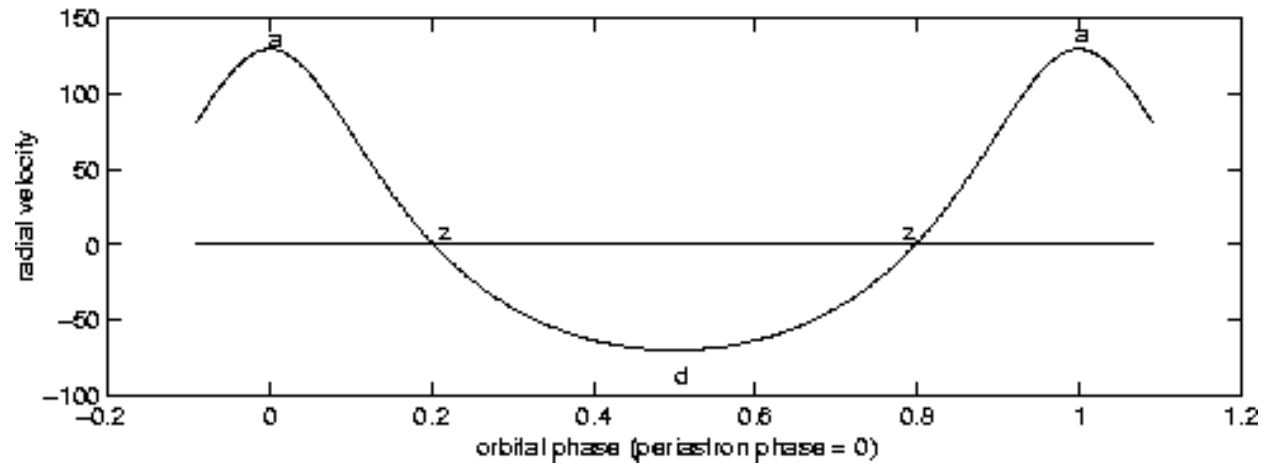


$$e = 0.1, w = 45^\circ$$

Radial Velocities

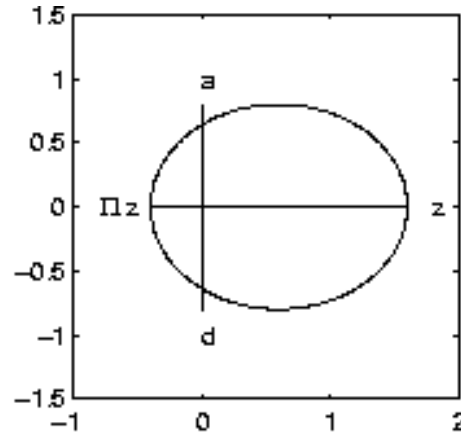


Less time near periastron

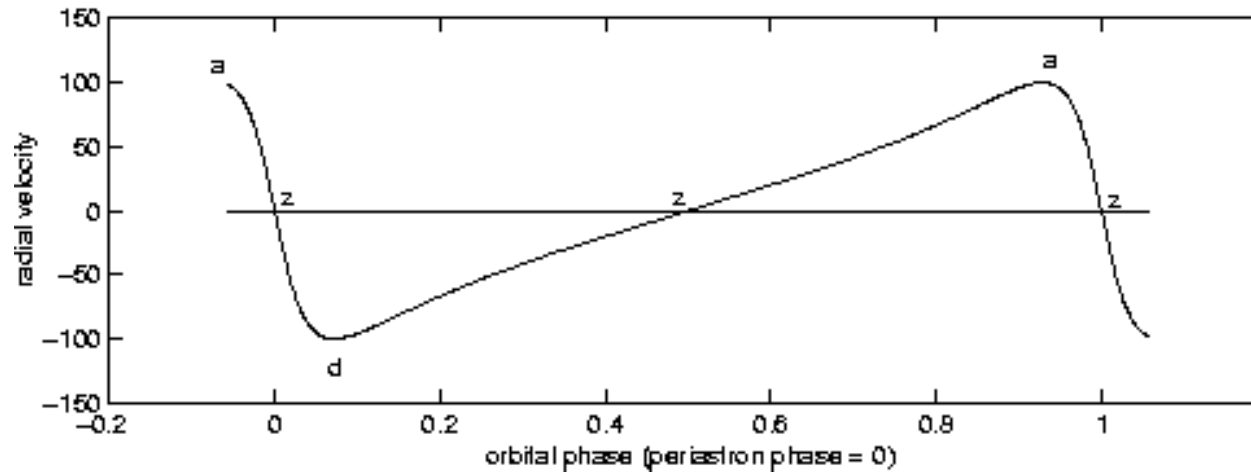


$$e = 0.3, w = 0^\circ$$

Radial Velocities

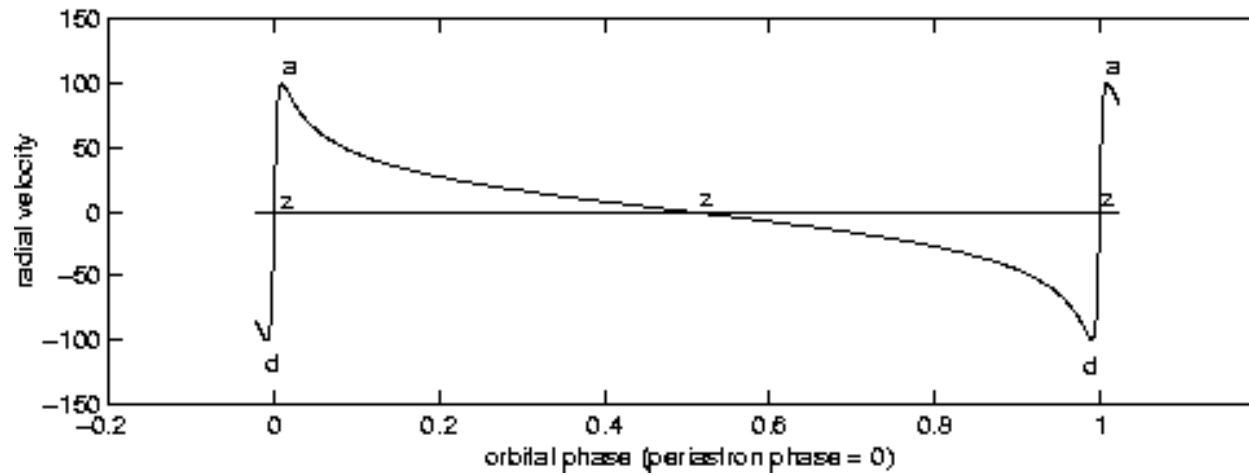
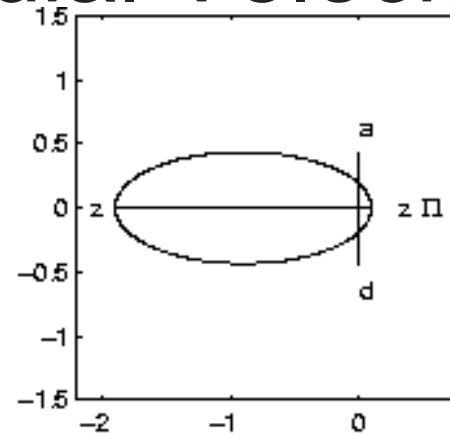


Max /Min velocity at ascending/descending nodes



- $e=0.6$, $w=90^\circ$

Radial Velocities



- $e=0.9$, $w=270^\circ$

Orbits from radial velocities

- **Observations** n radial velocity measurements:
 - $V_r(i)$ at times $t(i)$ $i = 1 \dots n$

- **Elliptical orbit model:**

$$E - e \sin E = \frac{2p}{P} (t - T)$$

$$\tan\left(\frac{\mathbf{q}}{2}\right) = \left(\frac{1+e}{1-e}\right)^{1/2} \tan\left(\frac{E}{2}\right)$$

$$V_r = K (\cos(\mathbf{q} + \mathbf{w}) + e \cos \mathbf{w}) + \mathbf{l}$$

- **Best fit determines** $(K, e, \mathbf{w}, \mathbf{g}T, P)$
- **e.g. using a least squares procedure**

Minimum masses

SB2 : measure P, K_1, K_2, e

$$K_i = \frac{2\mathbf{p} a_i \sin i}{P\sqrt{1-e^2}} \rightarrow a_i \sin i = \frac{\sqrt{1-e^2}}{2\mathbf{p}} K_i P$$

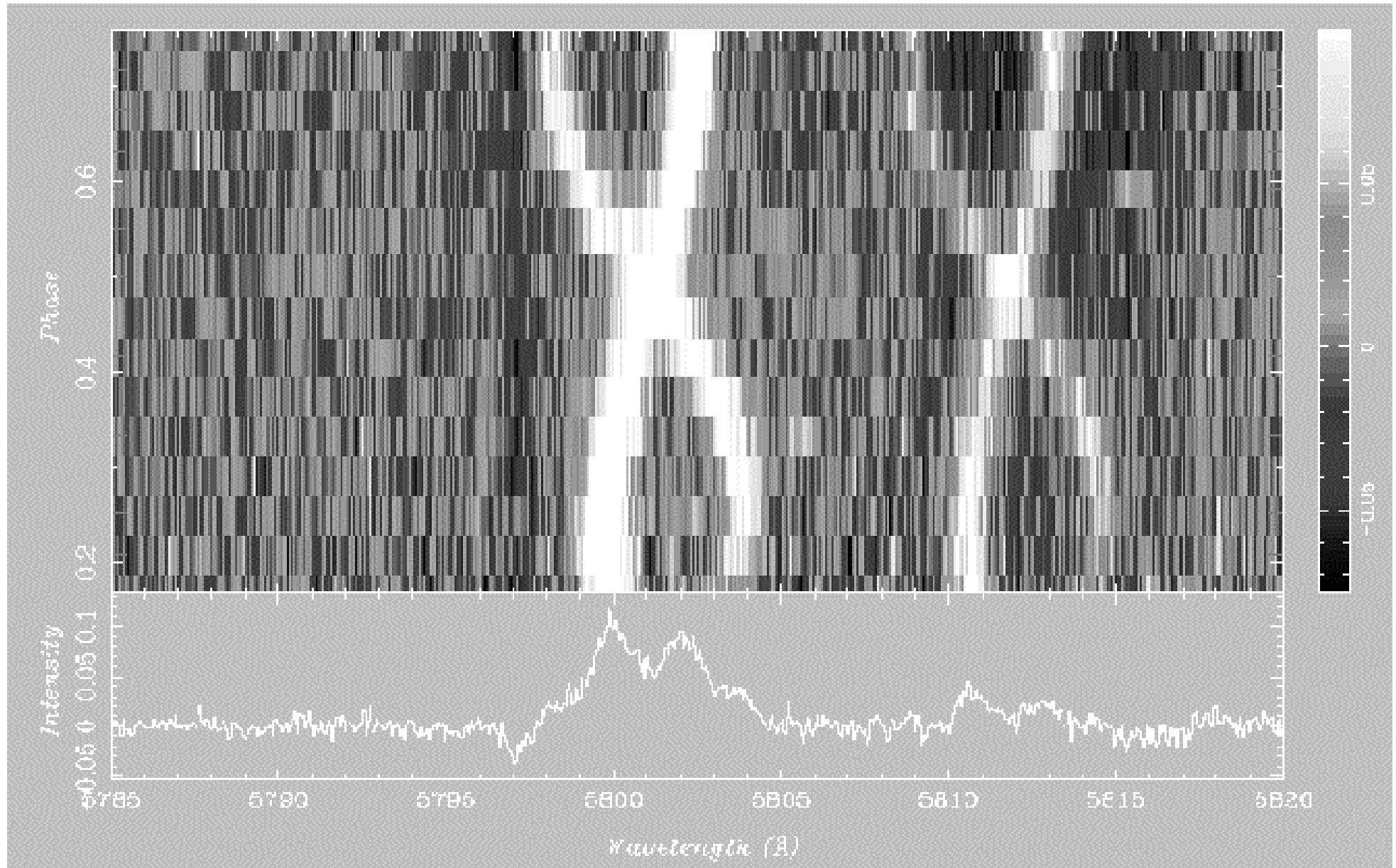
$$m_1 = M \frac{a_2}{a} = \frac{4\mathbf{p}^2 a^2 a_2}{G P^2}$$

$$m_1 \sin^3 i = \frac{4\mathbf{p}^2 a^2 a_2 \sin^3 i}{G P^2} = \frac{(1-e^2)^{3/2} (K_1 + K_2)^2 K_2 P}{2\mathbf{p} G}$$

SB1 : measure P, K_1, e , calculate the mass function :

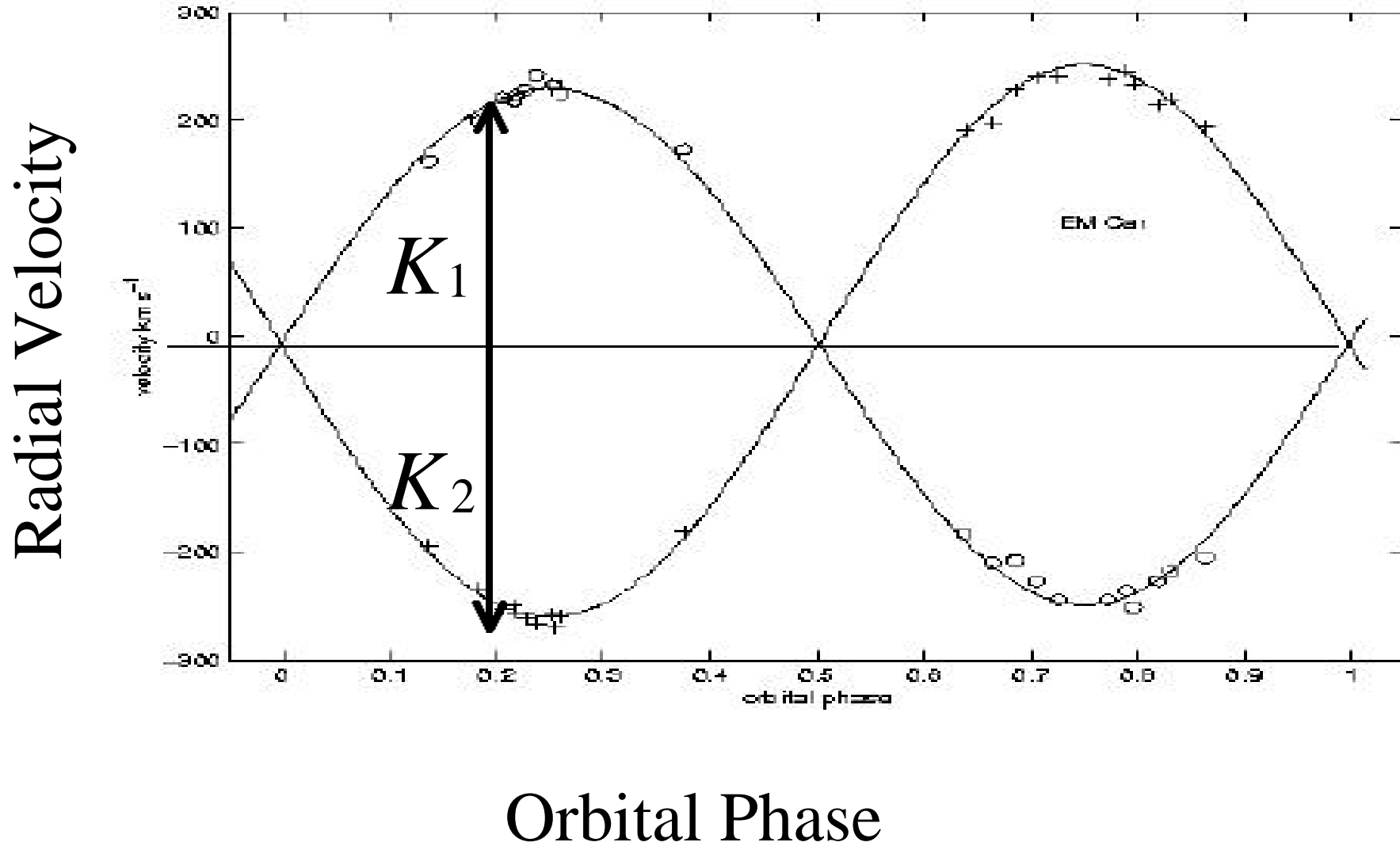
$$f(m_2) = \frac{m_2^3 \sin^3 i}{M^2} = \frac{(1-e^2)^{3/2} K_1^3 P}{2\mathbf{p} G}$$

KV Vel (sdO+M)

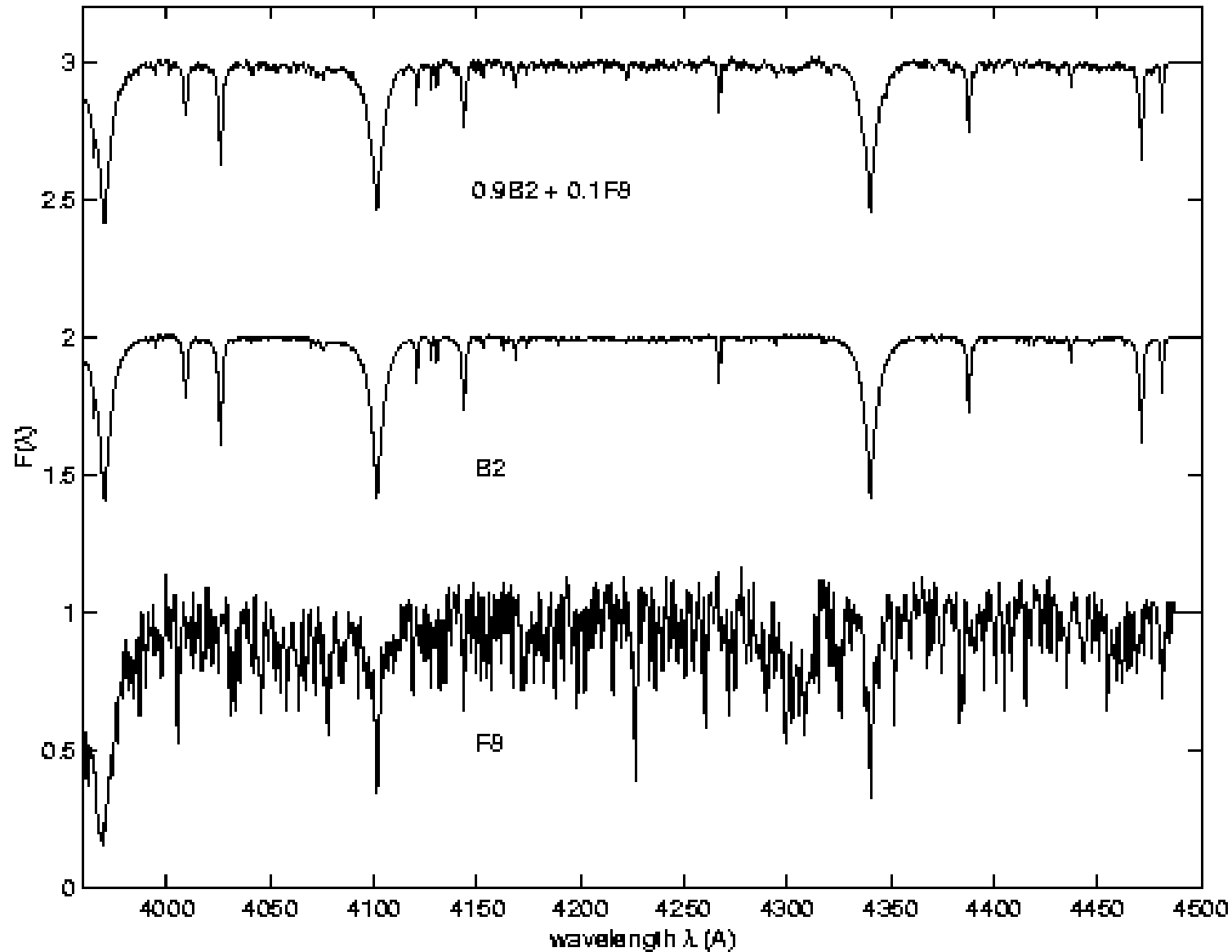


EM Car (O8V+O8V)

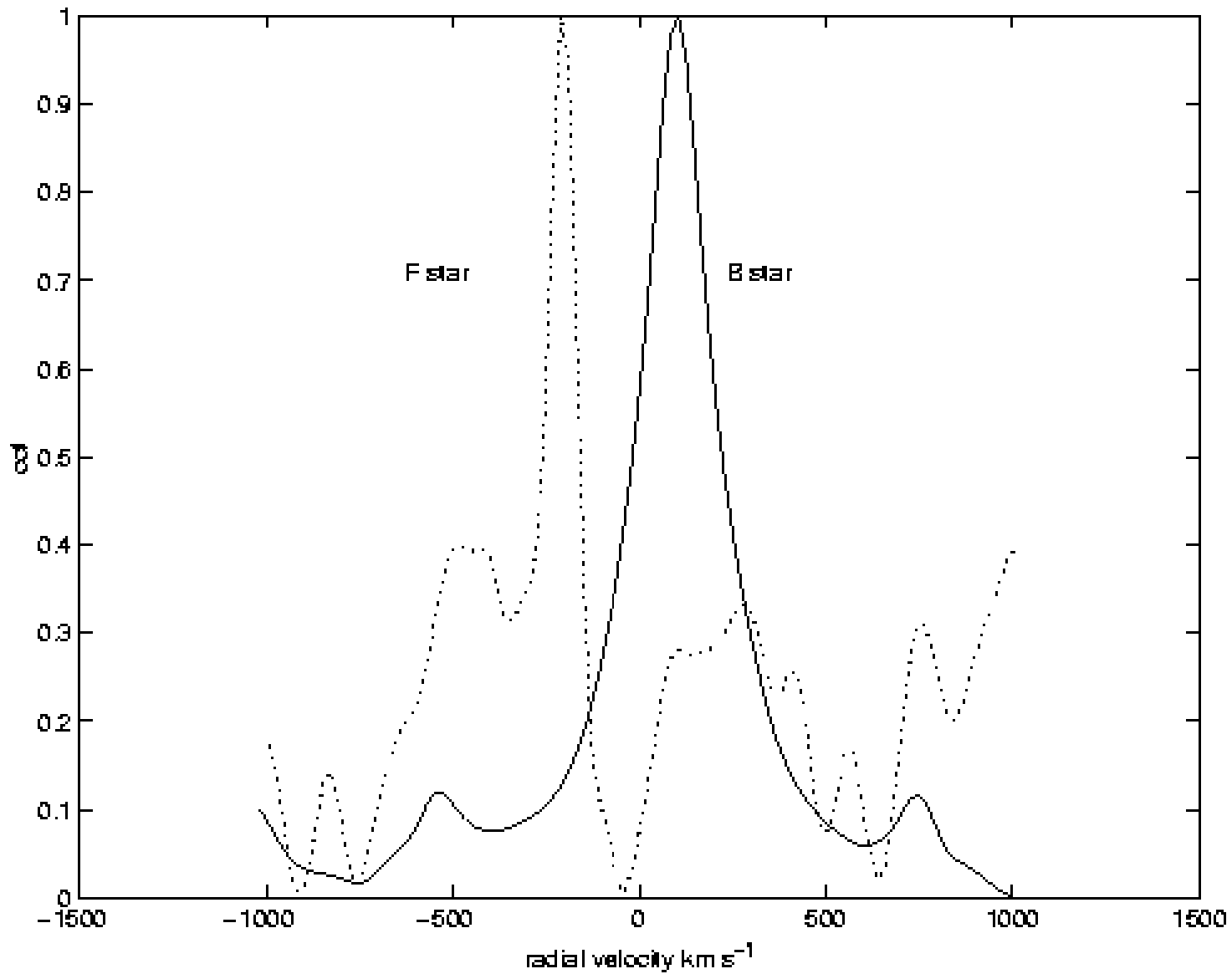
observe : $K = V \sin i$



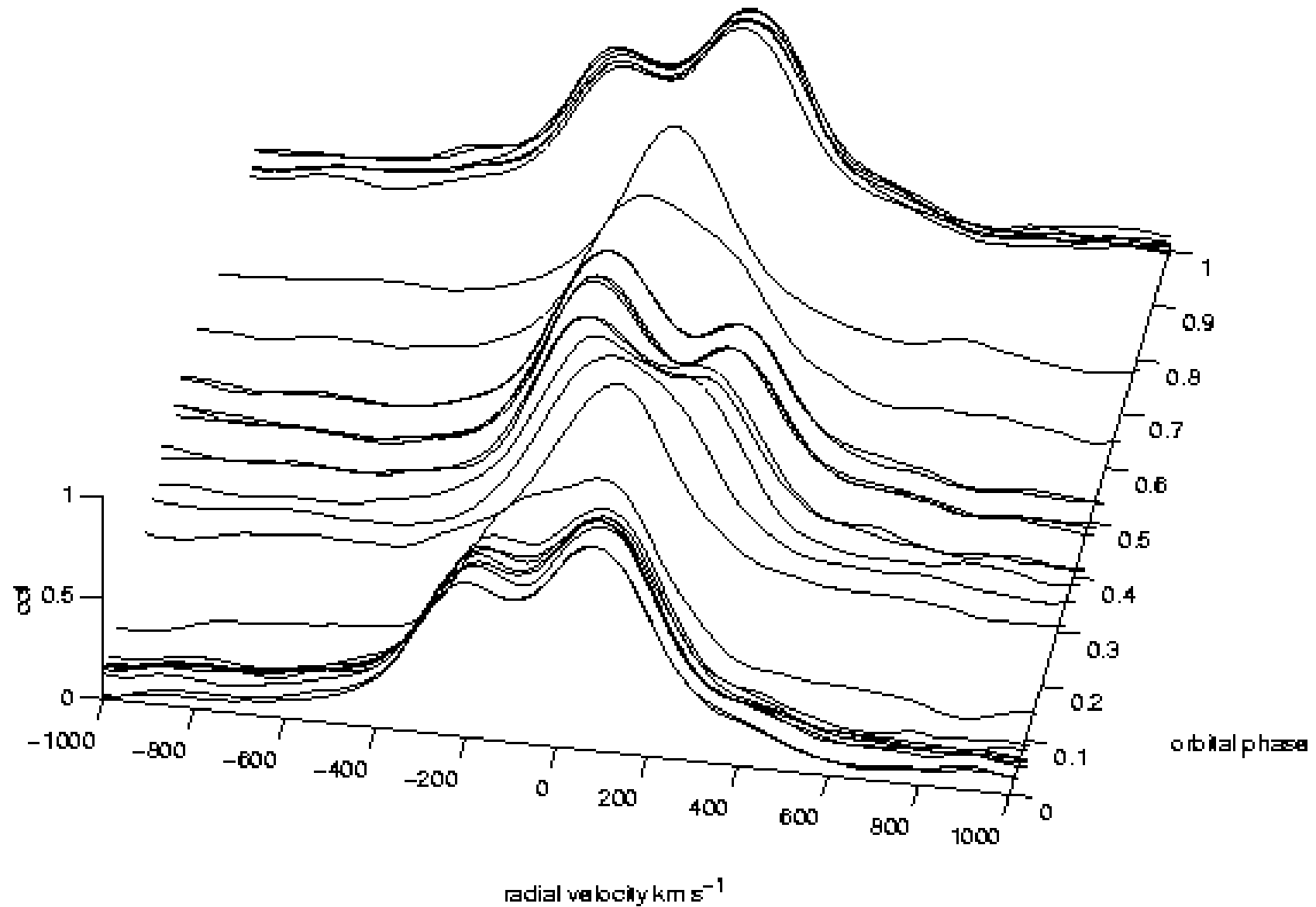
Composite spectrum (90% B2 + 10% F8)



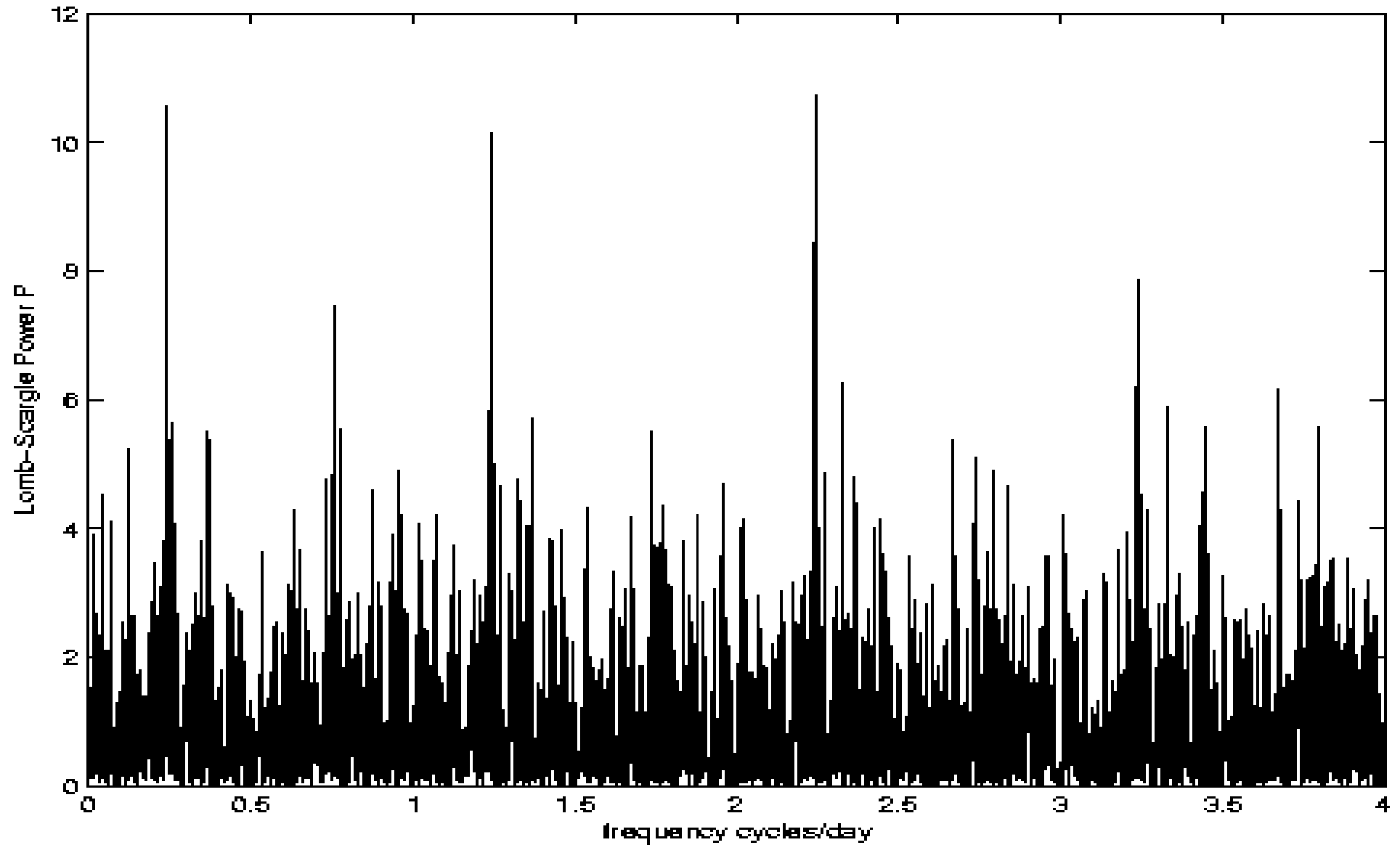
Cross-correlate with B2 and F8 template spectra



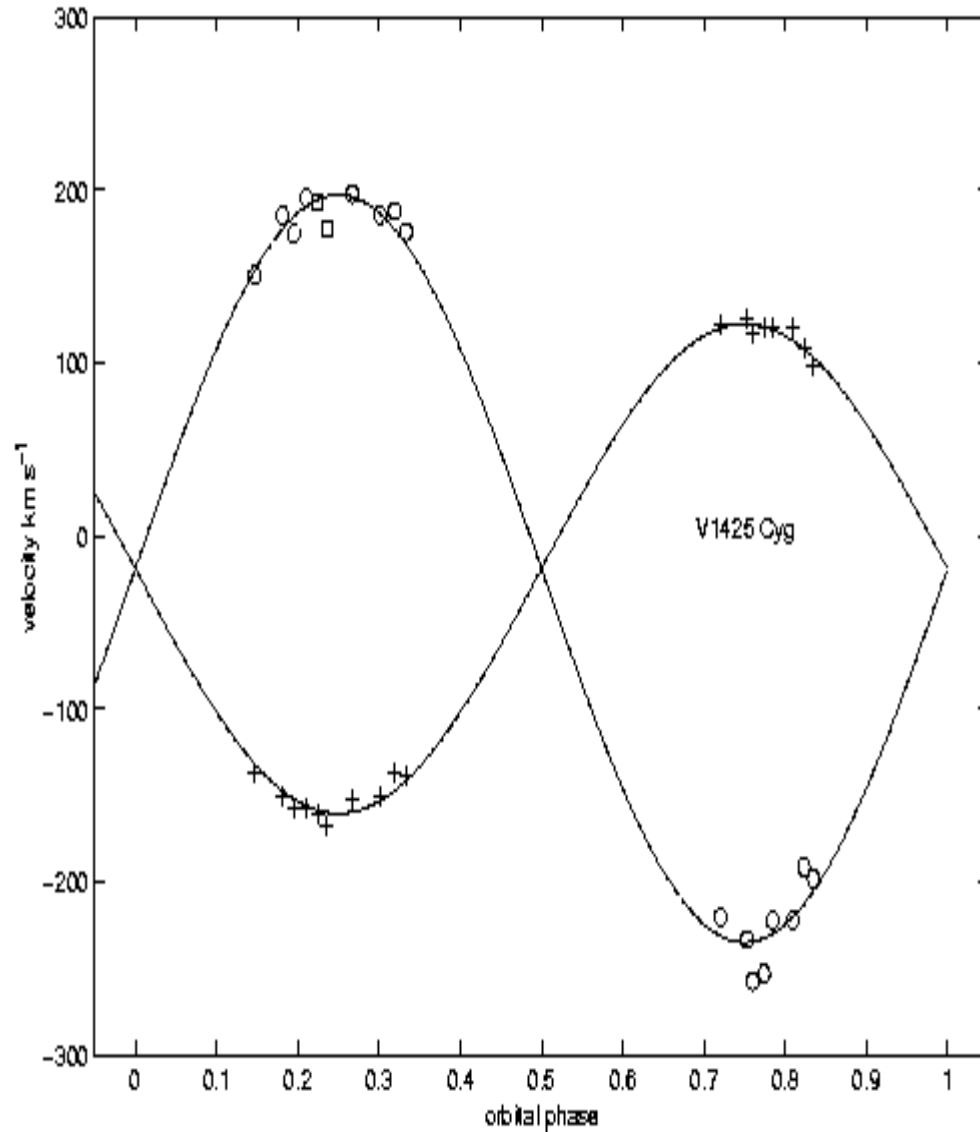
LZ Cep (O9.5V + O9.5V)



HR 1165 Periodogram

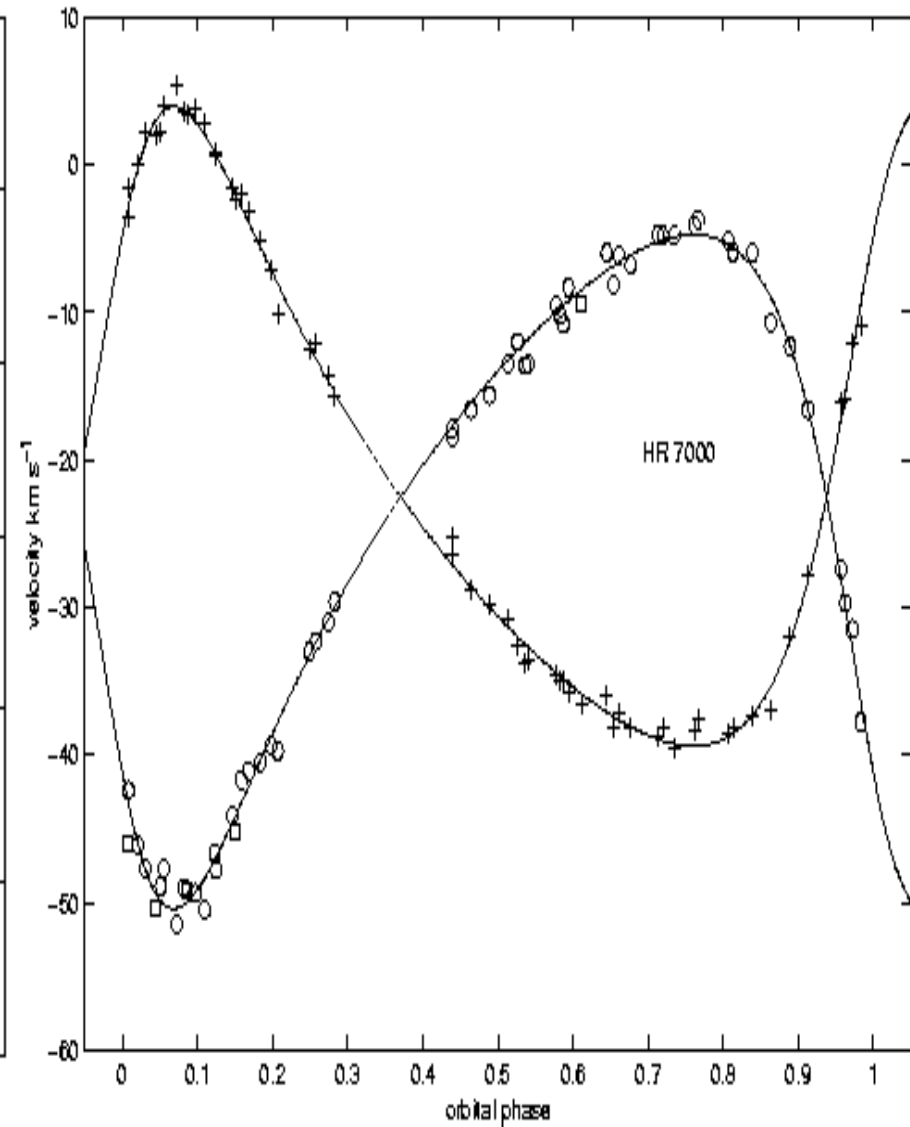


V1425 Cyg (B5V+B9V)



AS 4024

HR 7000 (F4V+ F6V)



Binary Stars and Accretion Disks