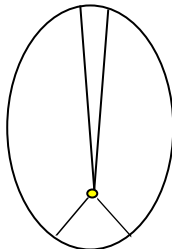


Kepler -- kinematics

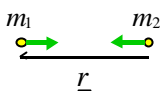
- Empirical fit to Tycho Brahe's planet motion observations
- Kepler's laws:
 1. Elliptical orbits Sun at focus
 2. Equal areas in equal times
 3. $P^2 \propto a^3$



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Newton -- dynamics

$$\underline{F}_1 = -\underline{F}_2$$


$$\underline{F} = m \ddot{\underline{r}}$$

$$\underline{F}_1 = -\frac{G m_1 m_2}{r^2} \hat{\underline{r}}$$

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The N-body problem

$$\Phi(\underline{r}, t) = -\sum_{i=1}^N \frac{G m_i}{|\underline{r} - \underline{r}_i|} = \text{gravitational potential}$$

$$\underline{f} = -\nabla\Phi = \text{acceleration}$$

$$\text{equation of motion: } m \ddot{\underline{r}} = \underline{F} = m \underline{f} = -m \nabla\Phi$$

$$\underline{J} \equiv \underline{r} \times (m \dot{\underline{r}}) = \text{angular momentum}$$

$$\dot{\underline{J}} = \underline{r} \times (m \ddot{\underline{r}}) + \dot{\underline{r}} \times (m \dot{\underline{r}})$$

$$= \underline{r} \times \underline{F} \equiv \underline{G} = \text{torque}$$

Note: If $\underline{F} \parallel \underline{r}$ then $\underline{J} = \text{constant}$

and $\underline{r}, \dot{\underline{r}}$ (both $\perp \underline{J}$) define orbital plane.

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Energy of orbit

$$\text{Kinetic Energy: } K \equiv \frac{1}{2} m |\dot{\underline{r}}|^2 = \frac{1}{2} m (\dot{\underline{r}} \cdot \dot{\underline{r}})$$

$$\dot{K} = m \dot{\underline{r}} \cdot \ddot{\underline{r}} = \dot{\underline{r}} \cdot \underline{F} = -m \dot{\underline{r}} \cdot \nabla\Phi$$

$$\text{Potential Energy: } W \equiv m \Phi$$

$$\dot{W} = m \dot{\Phi} + m \left(\frac{d\Phi}{dx} \frac{dx}{dt} + \frac{d\Phi}{dy} \frac{dy}{dt} + \frac{d\Phi}{dz} \frac{dz}{dt} \right)$$

$$= m \dot{\Phi} + m \nabla\Phi \cdot \dot{\underline{r}}$$

$$\text{Total Energy: } E = K + W$$

$$\dot{E} = \dot{K} + \dot{W} = (-m \dot{\underline{r}} \cdot \nabla\Phi) + (m \dot{\Phi} + m \nabla\Phi \cdot \dot{\underline{r}}) = m \dot{\Phi}$$

\therefore Energy is constant if $\dot{\Phi} = 0$

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Orbits in spherical potentials

If $\Phi(r)$ depends on $|\underline{r}|$, not \underline{r} , then

$$\underline{f} = -\nabla\Phi = -\frac{d\Phi}{dr} \hat{\underline{r}}$$

then

$$\dot{\underline{J}} = \underline{r} \times m \underline{f} = -m \frac{d\Phi}{dr} \underline{r} \times \hat{\underline{r}} = 0$$

\rightarrow angular momentum about origin is conserved !

\therefore orbital motion is in a plane

$\underline{r}, \dot{\underline{r}}$ in plane,

specific angular momentum

$$\underline{L} \equiv \frac{\underline{J}}{m} \text{ perpendicular to plane}$$

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Kinematics

position $\underline{r}(t)$

$$\text{velocity } \dot{\underline{r}} = \dot{r} \hat{\underline{r}} + r \dot{\underline{q}} \hat{\underline{q}}$$

$$\text{speed } V = \dot{r}^2 + r^2 \dot{\underline{q}}^2$$

unit vectors rotate with \underline{q} along the trajectory

$$\frac{d\hat{\underline{r}}}{dt} = \frac{d\hat{\underline{r}}}{d\underline{q}} \frac{d\underline{q}}{dt} = \dot{\underline{q}} \hat{\underline{q}}$$

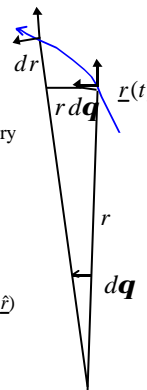
$$\frac{d\hat{\underline{q}}}{dt} = \frac{d\hat{\underline{q}}}{d\underline{q}} \frac{d\underline{q}}{dt} = -\hat{\underline{r}} \dot{\underline{q}}$$

acceleration

$$\ddot{\underline{r}} = \ddot{r} \hat{\underline{r}} + \dot{r} (\dot{\underline{q}} \hat{\underline{q}}) + \dot{r} \dot{\underline{q}} \hat{\underline{q}} + r \ddot{\underline{q}} \hat{\underline{q}} + r \dot{\underline{q}} (-\dot{\underline{q}} \hat{\underline{r}})$$

$$= (\ddot{r} - r \dot{\underline{q}}^2) \hat{\underline{r}} + (2 \dot{r} \dot{\underline{q}} + r \ddot{\underline{q}}) \hat{\underline{q}}$$

should be 0 if \underline{F} parallel to \underline{r}



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Kepler's 2nd Law

area swept out by \mathcal{L} in time dt

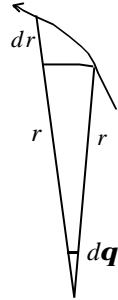
$$dA = \frac{1}{2}(r^2 d\mathbf{q} + r dr d\mathbf{q})$$

$$\dot{A} = \frac{1}{2} r^2 \dot{\mathbf{q}}$$

$$= \frac{1}{2} \mathcal{L} \times \dot{\mathcal{L}} = \frac{J}{2m} = \text{constant}$$

Note: $\frac{1}{r} \frac{d}{dt}(r^2 \dot{\mathbf{q}}) = 2\dot{r}\dot{\mathbf{q}} + r\ddot{\mathbf{q}} = 0$

transverse component of acceleration = 0



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Conic Sections

$$r = \frac{\ell}{1 + e \cos \mathbf{q}}$$

e = eccentricity

> 1 hyperbola

$= 1$ parabola

< 1 ellipse

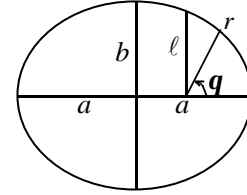
$= 0$ circle

ℓ = semi-latusrectum

a = semi-major axis

b = semi-minor axis

$$= a(1 - e^2)^{1/2}$$



$$\mathbf{q} = 0 \quad r = \frac{\ell}{1 + e} = a(1 - e) \quad \text{periastron}$$

$$\mathbf{q} = 90^\circ \quad r = \ell = a(1 - e^2)$$

$$\mathbf{q} = 180^\circ \quad r = \frac{\ell}{1 - e} = a(1 + e) \quad \text{apastron}$$

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